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## Iterative-Based Optimal-Inverse Feedforward for Output-Tracking of Nonminimum-Phase Systems

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### Abstract

Precision positioning plays a significant role in nonminimum-phase systems, especially toward biomedical applications. Precision control for pre-specified output-trajectory can be obtained by iteratively modifying the control input based on previous cycle data of the output-tracking error. This paper presents the implementation of iterative control together with optimal inversion-based feedforward that further improves the output-tracking performance of nonminimum-phase systems. Experimental results for the piezo-based flexible structure system are provided to illustrate the improvements.

**Keywords:** Iterative control, Feedforward control, Nonminimum-phase system.

### 1. Introduction

Precision positioning is important in biomedical applications, e.g., during high frequency imaging of soft samples such as cells using Atomic Force Microscopes (AFMs) [1]. Several control approaches have been developed to achieve high precision control of the positioning system [2]. The main contribution of this work is to investigate the use of iterative control together with optimal inversion-based feedforward.

Optimal inversion-based feedforward is one of the control approaches that can be used to achieve high output-tracking performance [2]. However, this method requires perfect knowledge of the system to obtain the exact tracking of the given output trajectory, i.e., perfect model of the system  $G$  and perfect measurement of the output  $y$  are necessary for the exact output-tracking. As a result, output-tracking errors due to incomplete information of the system and measurement could occur in practical applications.

In the previous study, the iterative inversion-based control method was shown to successfully reduce output-tracking errors by modifying the control input based on previous cycle data of the output-tracking error [3]. The optimal inverse  $G_{opt}^{-1}$  is used in this work for updating the control input in the iteration control law, instead of using the model inverse  $G^{-1}$  as in [3]. Experimental results for the nonminimum-phase piezo-based flexible structure system are provided to illustrate output-tracking improvements by using the iterative-based optimal-inverse feedforward approach. To summarize, the experimental magnitude of the output-tracking error can be reduced from 60% (using optimal inversion-based control without iteration) to 12% after five iteration steps.

### 2. Problem Formulation

#### 2.1 System description

Let the system ( $G$ ) be a linear time-invariant system

$$G(s) = \frac{y(s)}{u(s)} \quad (1)$$

where  $u(s)$  is the input of the system and  $y(s)$  is the output of the system in the Laplace domain.

#### 2.2 Control and positioning problems

The control problem is to find the input  $u$  such that the system output  $y$  tracks the desired output  $y_d$ . The output-tracking error  $e$  is defined by

$$e(t) = y(t) - y_d(t), \quad \forall t \geq 0. \quad (2)$$

The diagram of the output-tracking system is shown in Fig. 1.

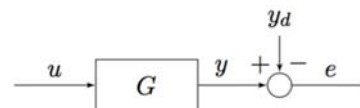


Fig. 1 Diagram of the output-tracking system.

A flexible beam is used for the positioning problem. In this work, the system model  $G$  is approximated by using the frequency response of the system. The input  $u$  is applied at one end and the output  $y$  is the displacement of the other end as shown in Fig. 2. Consequently, the example system is nonminimum-phase due to non-collocation of the input  $u$  and output  $y$  of the system.

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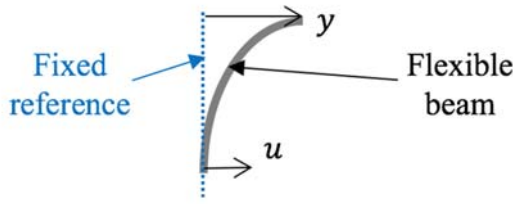


Fig. 2 A flexible beam is used in the positioning problem. The input  $u$  is applied at the bottom end and the output  $y$  is the displacement of the top end.

### 3. Control Approach

#### 3.1 Optimal inversion-based control

The frequency-weighted optimal inverse that minimizes the cost functional [4]

$$J(u) = \int_{-\infty}^{\infty} \{u^*(\omega)R(\omega)u(\omega) + e^*(\omega)Q(\omega)e(\omega)\} d\omega \quad (3)$$

where  $\omega$  is frequency,  $*$  denotes complex conjugate transpose, and positioning error  $e$  can be given by [2]

$$u(\omega) = \frac{G^*(\omega)Q(\omega)}{R(\omega) + G^*(\omega)Q(\omega)G(\omega)} y_d(\omega) = G_{opt}^{-1}(\omega) y_d(\omega) \quad (4)$$

and real-valued frequency-dependent weightings  $R$  and  $Q$  that penalize the size of the input  $u$  and the positioning error  $e$  are given by

$$R(\omega) = r^*(\omega)r(\omega), \quad Q(\omega) = q^*(\omega)q(\omega) \quad (5)$$

where values of  $r$  and  $q$  can be chosen to suppress high frequencies based on the frequency of the desired output-trajectory. In practical applications, perfect model of the system  $G$  and perfect measurement of the output  $y$  may not be possible to obtain, due to several factors such as nonlinearities of the system and sensor noise. Experimental output-tracking errors generally occur with the use of optimal inversion-based control alone.

#### 3.2 Iterative-based optimal-inverse control

Output-tracking performance can be improved for periodic trajectory, for example, scanning of the scanning probe microscopy (SPM) probe, by using iterative control approaches [2]. In this work, the optimal inverse system  $G_{opt}^{-1}$  is used in the iteration law as

$$u_{k+1}(\omega) = u_k(\omega) + \rho(\omega)G_{opt}^{-1}(\omega)[y_d(\omega) - y_k(\omega)] \quad (6)$$

where the iteration step  $k \geq 1$  and the frequency-dependent iteration gain  $\rho$  can be chosen based on criteria given in [3] which is also summarized in this section. The difference  $\Delta_G$  between the system model obtained from frequency response data and the optimal inverse are given by

$$\Delta_G(\omega) = \frac{G(\omega)}{G_{opt}(\omega)} = G(\omega)G_{opt}^{-1}(\omega) = \Delta_r(\omega)e^{j\Delta_\theta(\omega)} \quad (7)$$

where  $\Delta_r$  is the magnitude difference and  $\Delta_\theta$  is the phase difference. The iteration gain  $\rho$  can be chosen by

$$0 < \rho(\omega) < \frac{2 \cos(\Delta_\theta(\omega))}{\Delta_r(\omega)} \quad (8)$$

and the magnitude of the phase difference  $\Delta_\theta$  is less than  $\pi/2$

$$|\Delta_\theta(\omega)| < \pi/2 \quad (9)$$

## 4. Results

#### 4.1 Experimental system

The experiment system consists of a piezo-based flexible structure where the displacement of the tip of the flexible beam was measured using an inductive sensor and all electrical signals were controlled using data acquisition (DAQ) as shown in Fig. 3.

A flexible beam shown in Fig. 2 was made of a thin brass sheet with dimensions of 10 mm (width) by 25 mm (length) and 0.127 mm (thickness). As shown in Fig. 4 (bottom right), the one end of the flexible beam was attached to a piezo-actuator (Piezosystem Jena PA 16/14SC EXT) and the displacement of the other end was measured by an inductive sensor (Kaman SMU-9000-15N). A bi-polar power amplifier (Fig. 4, bottom left) with the gain of two was used to drive the piezo-actuator. The National Instruments NI USB-6211, see Fig. 4 (top right), was used to control electrical signals with its analog input and output channels.

#### 4.2 System model

The system  $G$  in Eq. (1) can be experimentally obtained by using the frequency response of the system, i.e., using the diagram given in Fig. 3

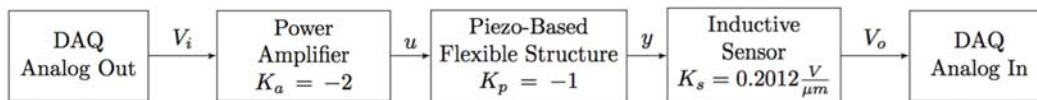


Fig. 3 Experimental system diagram

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$$G(\omega) = \frac{y(\omega) (\mu\text{m})}{u(\omega) (V)} = \frac{V_o(\omega)/K_s}{K_a K_p V_i(\omega)} = \left( \frac{1}{K_s K_a K_p} \right) \left( \frac{V_o(\omega)}{V_i(\omega)} \right) \quad (10)$$

where  $\omega$  is frequency ( $\text{rad/s}$ ),  $V_i$  is the pre-amplified input voltage ( $V$ ),  $V_o$  is the output voltage from the inductive sensor ( $V$ ),  $K_s = 0.2012 V/\mu\text{m}$  is the inductive sensor gain,  $K_a = -2$  is the power amplifier gain, and  $K_p = -1$  is the piezo-actuator gain. For the experimental system, the input  $u$  is the voltage applied to the piezo-actuator in Volts ( $V$ ) and the output  $y$  is the displacement of the tip of the flexible beam in microns ( $\mu\text{m}$ ).

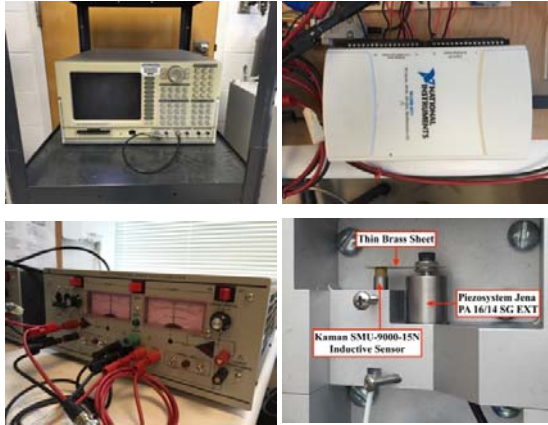


Fig. 4 Experimental system: (top left) dynamic signal analyzer (SR785), (top right) data acquisition (DAQ), (bottom left) power amplifier, and (bottom right) piezo-based flexible structure with inductive sensor (Kaman SMU-9000-15N)

Figure 5 shows frequency response of the piezo-based flexible system obtained from a dynamic signal analyzer (SR785), see Fig. 4 (top left) by sending sine wave signals at different frequencies (between 50 and 2,500  $\text{Hz}$ ) to the system and measuring the displacement of the tip of the flexible beam with the inductive sensor. As shown in Fig. 5, example nonminimum-phase zeros are at a frequency of 600  $\text{Hz}$ , approximately.

### 4.3 Desired output trajectory

A triangle trajectory, used in several scanning (such as AFM) operations [5], was used as the desired output-trajectory to illustrate the control approaches for the piezo-based flexible structure. One cycle data of the desired output-trajectory  $y_{d,i}$  are shown in Fig. 6 (top plot) and the complete (ten-cycle) data of the desired output-trajectory are shown in Fig. 6 (bottom left plot). Components of the desired output-trajectory at different frequencies obtained using Fourier transform are also shown in Fig. 6 (bottom right plot)

where the fundamental frequency is at 10  $\text{Hz}$ .

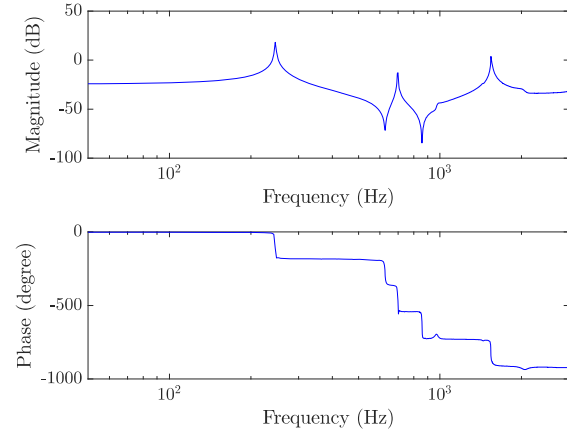


Fig. 5 Frequency response of the piezo-based flexible structure

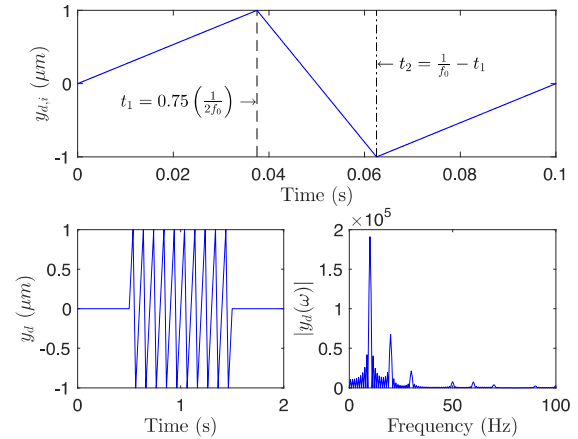


Fig. 6 Desired output-trajectory: (top) one-cycle data, (bottom left) ten-cycle data, and (bottom right) magnitude of the desired output-trajectory

### 4.4 Optimal inversion-based control

While the simulation result obtained by using optimal inversion-based control shows reasonable output-tracking result, experimental output-tracking errors appear to be large, as shown in Fig. 9 (bottom left plot). Based on the frequency of the desired output-trajectory, values of  $r$  and  $q$  are chosen to be

$$w(s) = \frac{2\pi 50}{s + 2\pi 50}, \quad q(s) = 1000 \times w(s), \quad r(s) = 1/w(s) \quad (11)$$

throughout this work. Magnitudes of the weighting matrices  $Q$  and  $R$  are shown in Fig. 7. The simulation result obtained using the optimal inversion-based control shows that the output  $y$  follows the desired trajectory, Fig. 8 (bottom plot). The experimental output-tracking results from the optimal inversion-based control, i.e., zero iteration step ( $k = 0$ ), are shown in Fig. 9 (left plots). The averaged data of seven

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trials of the output  $y$  were used to compute the output-tracking error  $e$ . Maximum output tracking error is  $0.60 \mu\text{m}$  which is equal to 60% of the amplitude of the desired output-trajectory ( $1.5 \mu\text{m}$ ) as shown in Fig. 12 (bottom plot) and the output-tracking error  $e$  with zero iteration step ( $k = 0$ ) is shown in Fig. 12 (top left plot).

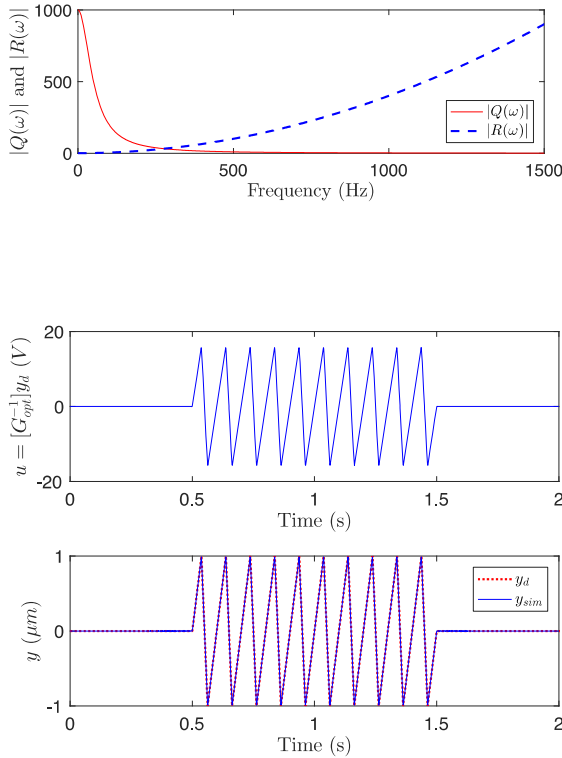


Fig. 8 Simulation result using optimal inversion-based control: (top) input and (bottom) output

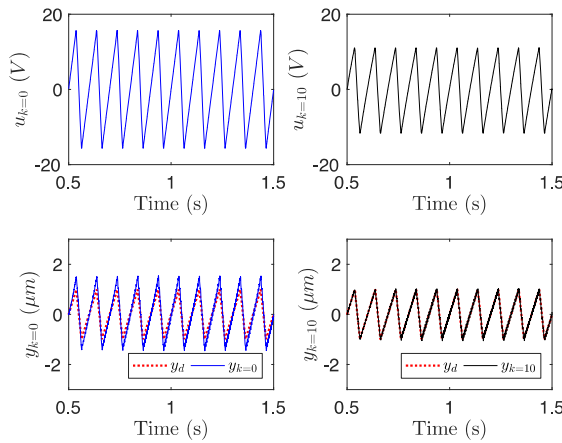


Fig. 9 Experimental data with and without iterative inversion-based control: (left) results without iteration ( $k = 0$ ), (right) results with iteration ( $k=10$ ), (top) input, and (bottom) output

### 4.5 Iterative-based optimal-inverse control

To decrease the output-tracking error, iterative control was used with the appropriate iteration gain  $\rho$  based on criteria given in Eqs. (8) and (9). The magnitude and phase analyses are shown in Fig. 10 (top and middle plots). As shown in Fig. 10 (bottom plot), the maximum allowable iteration gain  $\rho$  is two ( $\rho = 2$ ). Nevertheless, the iteration gain  $\rho$  used in the experiments was chosen to be

$$\rho(\omega) = \begin{cases} 1, & |\omega| \leq 2\pi 200 \\ 0, & \text{otherwise} \end{cases} \quad (12)$$

which was sufficiently large for convergence as shown in Fig 11. In addition, the control input were updated during time  $t \in [0.5, 1.5]$  s to neglect effects of noise from the inductive sensor. The experimental output-tracking results after ten iteration steps ( $k = 10$ ) are shown in Fig. 9 (right plots). Maximum output-tracking error during each iteration step is shown in Fig. 12 (bottom plot) and the output-tracking error  $e$  with zero iteration step ( $k = 10$ ) is shown in Fig. 12 (top right plot). Maximum output-tracking error after five iteration steps ( $k = 5$ ) is  $0.12 \mu\text{m}$  which is equal to 12% of the amplitude of the desired output-trajectory. In summary, the maximum output-tracking error was reduced from 60% to 12% after five iteration steps.

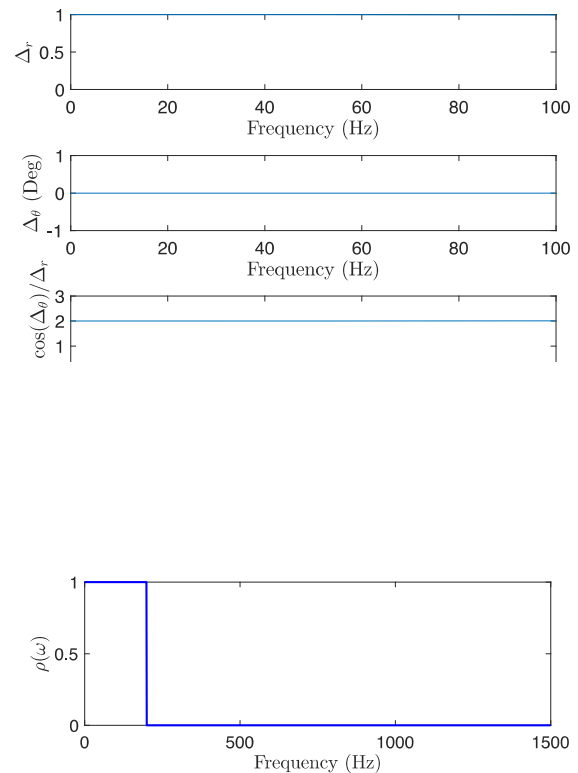
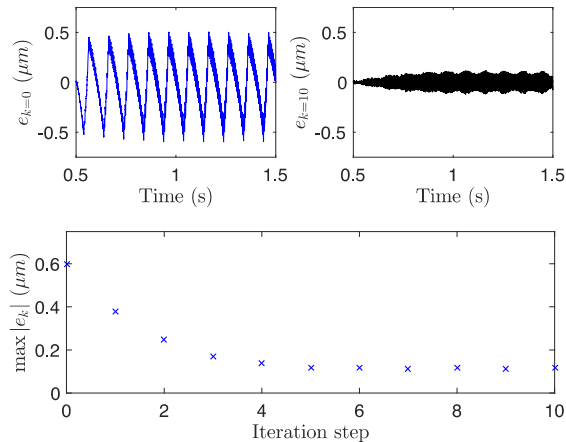


Fig. 11 Iteration gain used in experiments

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Fig. 12 Experimental output-tracking error: (top left) without iteration, (top right) without iteration, and (bottom) maximum output-tracking error at each iteration step

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